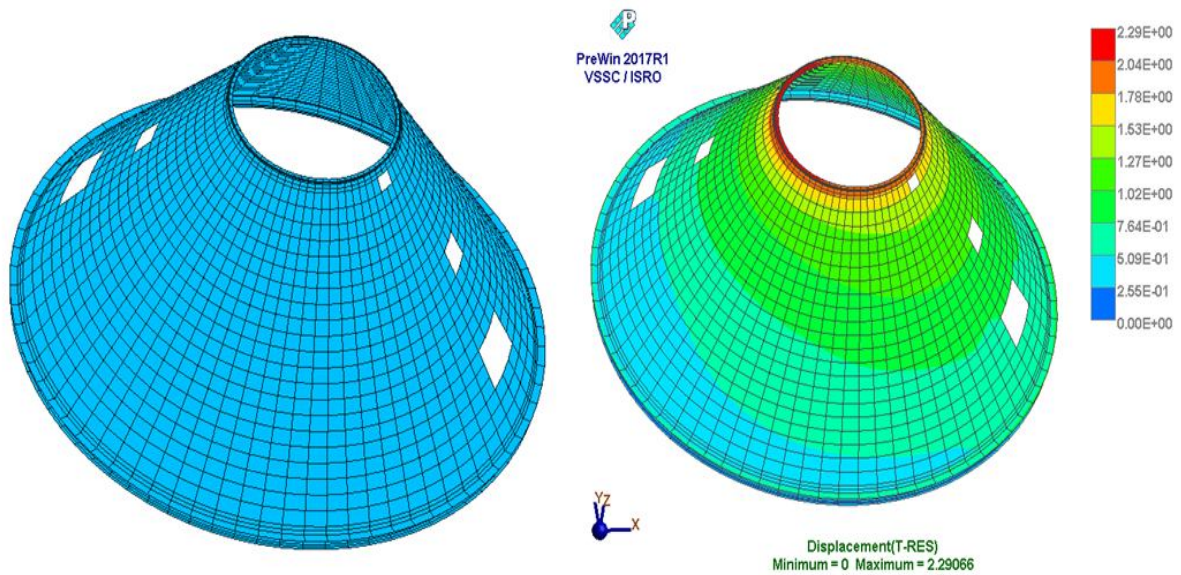


## Static analysis of typical pay load adaptor of a launch vehicle



**Composite structure**

**Materials used**

**Material property :**

1.  $E_{11} = 27500 \text{ MPa}$ ,  $E_{22} = 1600 \text{ MPa}$ ,  $G_{12} = G_{13} = G_{23} = 355 \text{ MPa}$ ,  $\nu = 0.155$ , Specific gravity = 1.7
2.  $E_{11} = 7000 \text{ MPa}$ ,  $E_{22} = 7000 \text{ MPa}$ ,  $G_{12} = G_{13} = G_{23} = 2900 \text{ MPa}$ ,  $\nu = 0.3$ , Specific gravity = 2.7
3.  $E_{11} = 0.00021 \text{ MPa}$ ,  $E_{22} = 0.00021 \text{ MPa}$ ,  $G_{12} = G_{13} = G_{23} = 0.0009 \text{ MPa}$ ,  $\nu = 0.3$
4.  $E_{11} = 1\text{E-}5 \text{ MPa}$ ,  $E_{22} = 1\text{E-}5 \text{ MPa}$ ,  $G_{12} = 1 \times 10^{-5} \text{ MPa}$ ,  $G_{13} = G_{23} = 39.3 \text{ MPa}$ ,  $\nu = 0.155$ , Specific gravity = 1.7

**Element types :** Quadrilateral shell

<b>Finite element statistics</b>	<b>Number of elements</b>	<b>Number of nodes</b>	<b>Degrees of freedom</b>
:	2278	2871	17004

Output parameter	FEAST <sup>SMT</sup>	NISA <sup>®</sup>
Displacement, mm	2.29	2.45